



# **WISH IFS DESIGN CONSIDERATIONS**

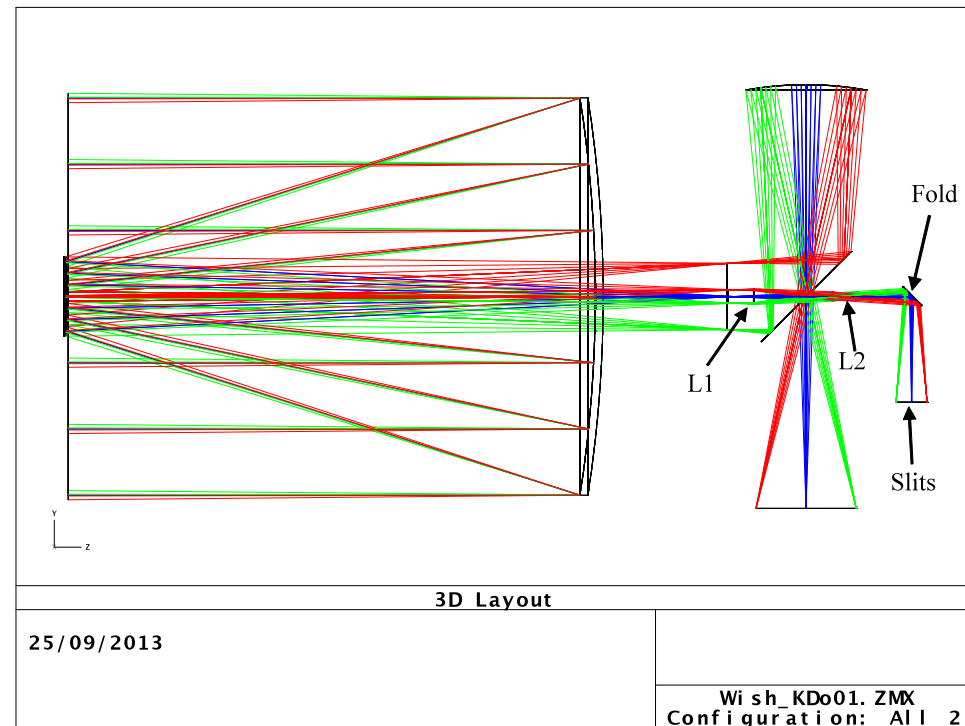
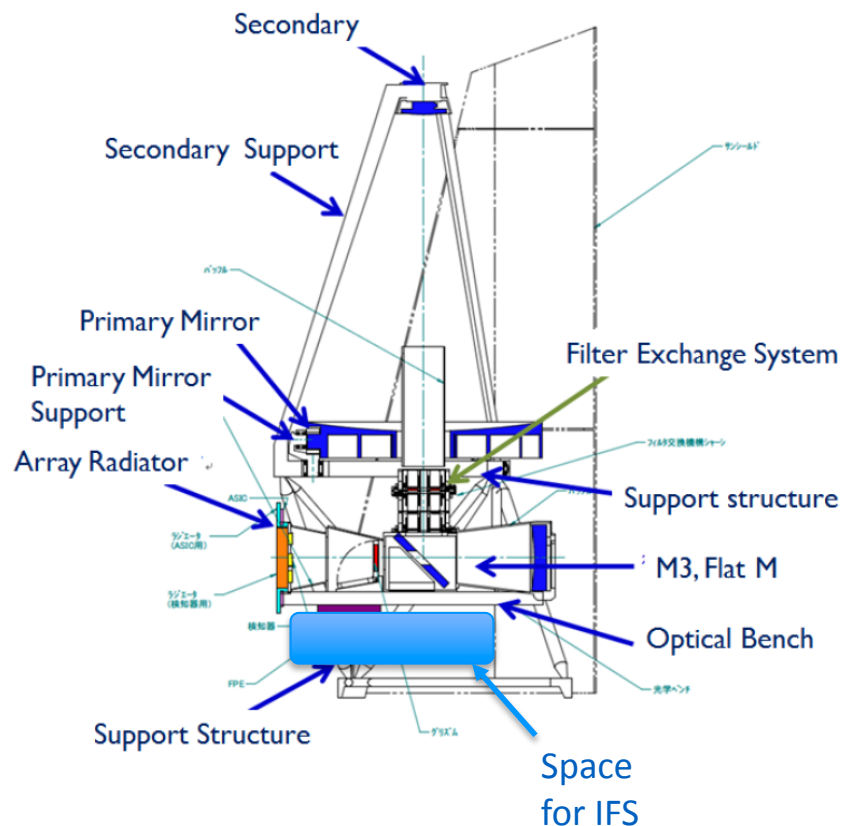
**Sebastien Vives, Kjetil Dohlen, Denis Burgarella**

**LAM, 11/9/14**

## Basic idea of implementation

**A small central FOV can be available through the hole in the fold mirror**

- ⦿ Possibly passing up to a few arc minutes towards the bottom of the space craft
- ⦿ Space available below the optical bench
- ⦿ A “reasonable” design could fit in a (large) shoebox



# Wanted main characteristics

## WISH list

- ⊙ Integral field spectrometer with FOV:  $1 \times 1'$
- ⊙ Angular resolution:  $0.5''$
- ⊙ Spectral range:  $1\text{-}5\mu\text{m}$  (at least  $2.2\text{-}5\mu\text{m}$ )
- ⊙  $R = \lambda/\Delta\lambda$ : 600 - 1000

## For comparison: JWST IFU characteristics

- ⊙ FOV:  $3 \times 3''$
- ⊙ Angular resolution:  $0.1''$  per slice, ie 30 slices
- ⊙ Spectral range:  $0.6\text{-}5\mu\text{m}$
- ⊙ R: 100, 1000, 2700

## Consequences:

- ⊙ For Nyquist sampling, need two slices per resolution element
- ⊙ Then we would need  $2 \times 1'/0.5'' = 240$  slices
- ⊙ A reasonable number is 40 slices, otherwise need multiple units, cf MUSE
- ⊙ Spectral range  $1\text{-}5\mu\text{m}$  covers more than two octaves, so would need three different gratings
- ⊙ For single-grating system should limit range to an octave, eg  $2.2\text{-}4.4$  or  $2.5\text{-}5$

# MIS-KIS approach (make-it-simple-keep-it-simple)

## System based on a single slicer stack and Hawaii-2RG detector

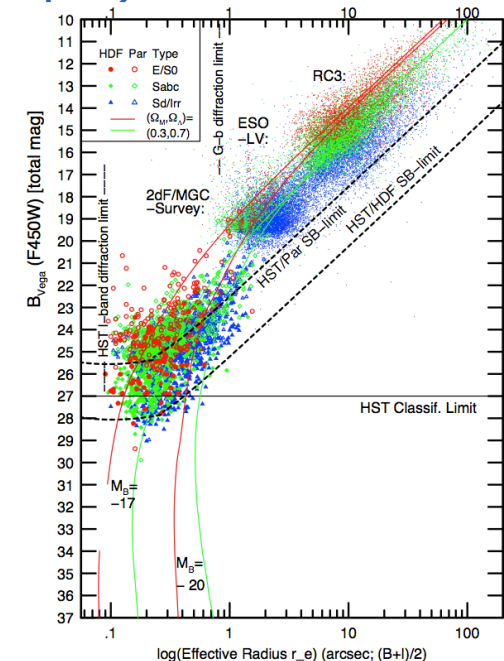
- ⊙ Assume 40 slices
- ⊙ Go to 30"
  - ⊙ still 10 x JWST's IFU on a side
  - ⊙  $\sim 100$  x JWST's IFU in area
- ⊙ Spatial sampling is then defined by the 0.75" slice width
- ⊙ For a typically 0.4" diameter galaxy this means forgetting about Nyquist and letting the object define the effective slit

## For spatial along-slice sampling equal to across-slice sampling

- ⊙ Need anamorphic fore optics (classical)
- ⊙  $40 \times 40 = 1600$  spatial detector pixels

## System based on a single grating

- ⊙ Spectral range limited to an octave, say 2.5-5 $\mu\text{m}$
- ⊙ Object defines the spectral resolution, assume 2 spectral pixels on an object of 0.4"
- ⊙ Spectral resolving power of  $R=1000$  requires  $\sim 2000$  spectral detector pixels



# Some scientific trades-off ...



The present design suggests that the MIS-KIS is  $\text{FOV} = 0.5'$ ,  $R = 1000$  and  $\Delta\lambda \approx 2.5 - 5 \mu\text{m}$  (1 octave).

- Is it possible (simple ?) to modify the design to reach  $\text{FOV} = 1.0'$

*Tentative answer:*

1) You can double the spatial sample size, ie  $1.5''$ , but then your camera will become more difficult to realize ( $f/D \sim 1.7$ )

2) You can split the field and have a set of 4 slicer stacks, spectrographs, cameras.

- Is it possible (simple ?) to modify the design to reach  $\Delta\lambda \approx 1.25 - 5 \mu\text{m}$  ?

*Tentative answer:*

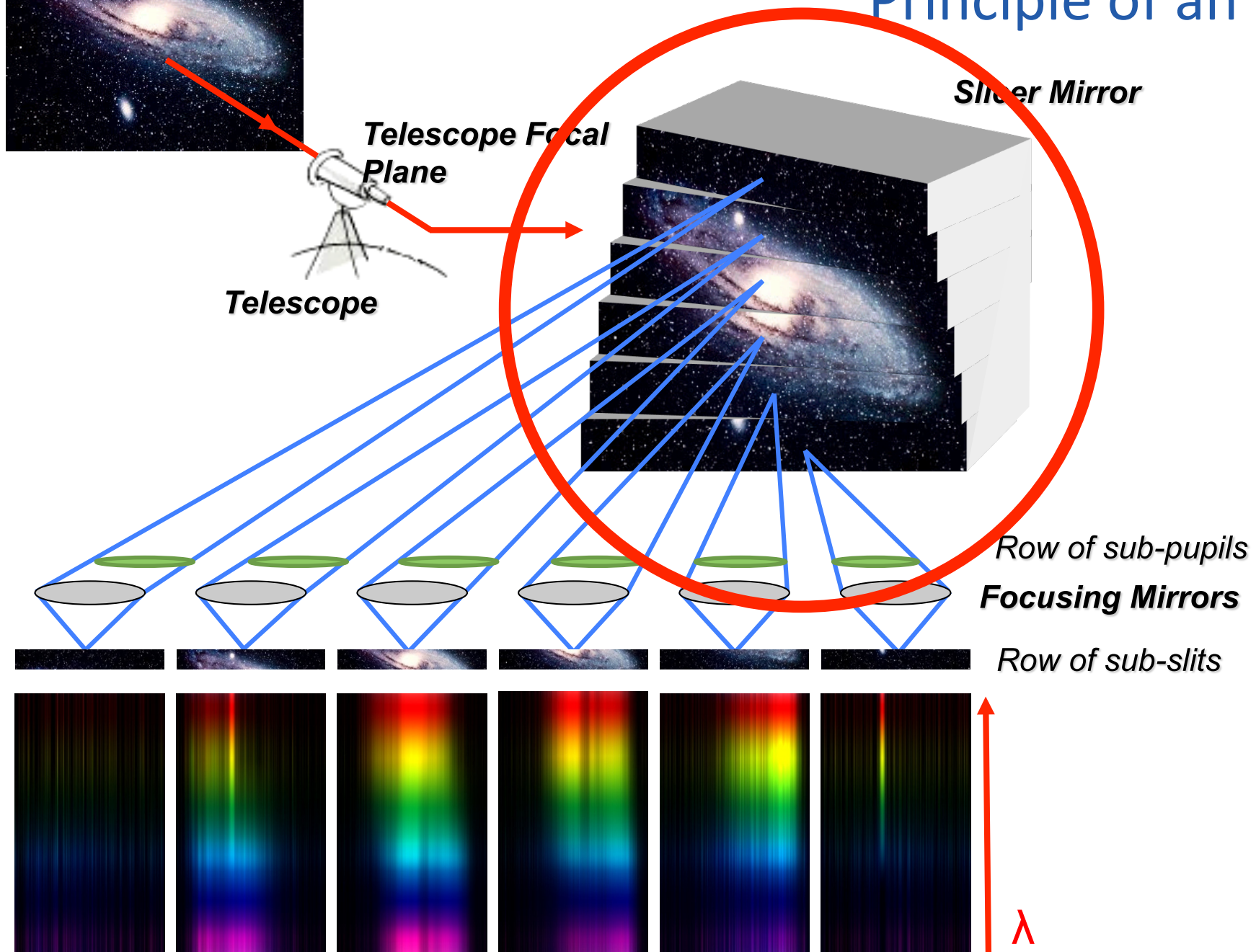
*This would require two gratings and could be obtained either using an exchange mechanism with a single spectrograph or by splitting the flux with a dichroic somewhere after the slicer stack and have two spectrographs, hence two detector arrays.*

- We might imagine to relax the constraint on  $R$  to  $R = 500 - 600$ , does that help?

*Hmm, help in what sense? It would liberate pixels, but it would not reduce complexity, since you then need to pipe the light from different spectrograph units onto a small detector chip.*

# LAM's heritage in Image slicer Instruments

# Principle of an IFS

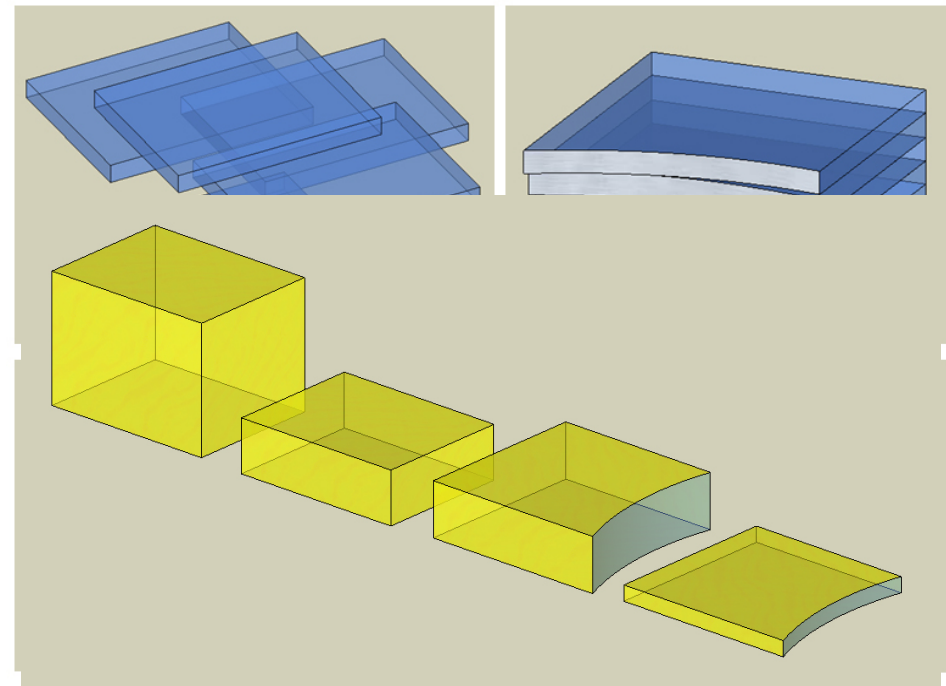
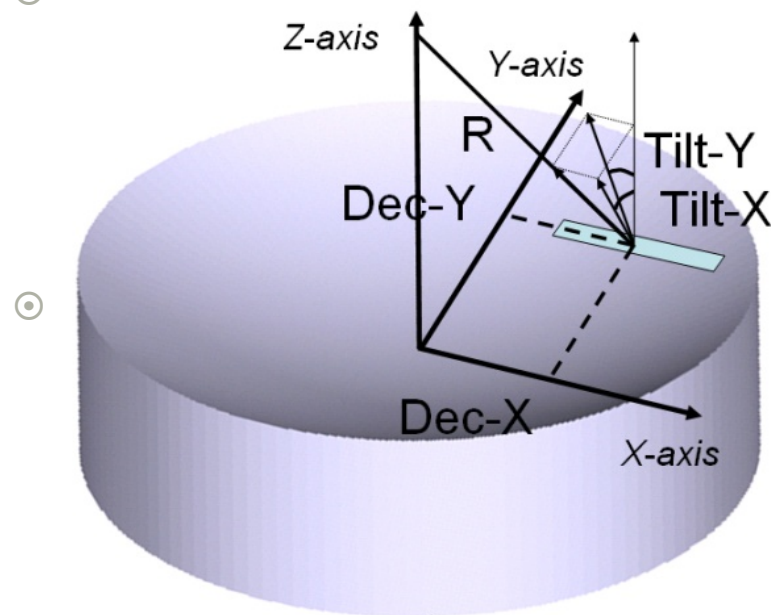




# High performance Low-cost IFU

**In 2006, LAM and Winlight proposed an innovative approach to make slicer mirrors**

- ⊙ A single classical polishing process to polish several dozen of slices.
- ⊙ Therefore we are able for the first

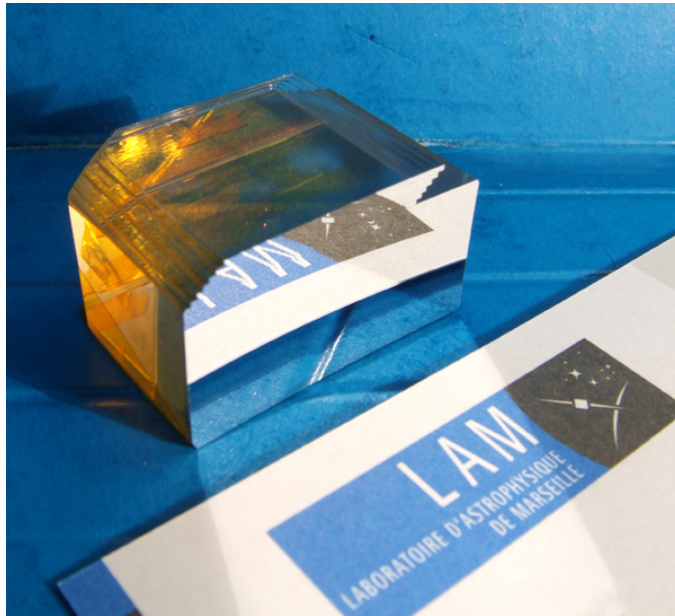


**Vives, S.; Prieto, E., Salaun, Y., Godefroy, P.,**

**“New technological developments in Integral Field Spectroscopy”, Proc. SPIE 7018, 70182N  
(2008)**



## First Demonstrator (2006)...



### **PERFORMANCE**

Generic optical specification for these components using our approach:

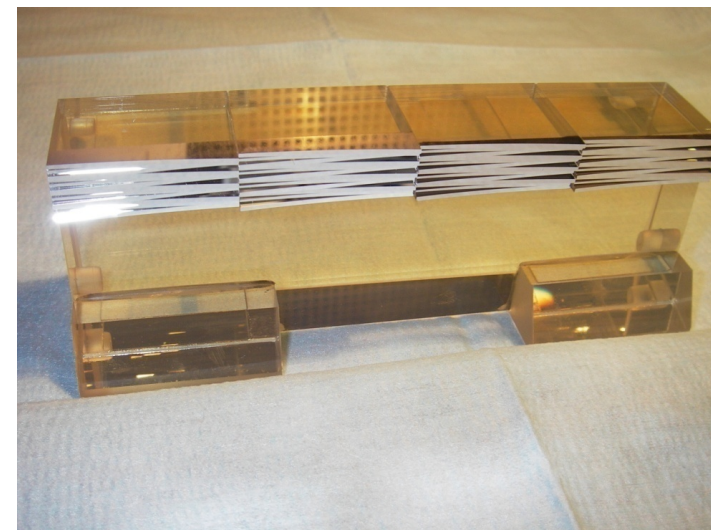
Surface quality  
 **$\sim \lambda/100$  rms**

Surface roughness  
 **$\sim 0.4$  nm**

Edges  
 **$\sim 1$   $\mu$ m**

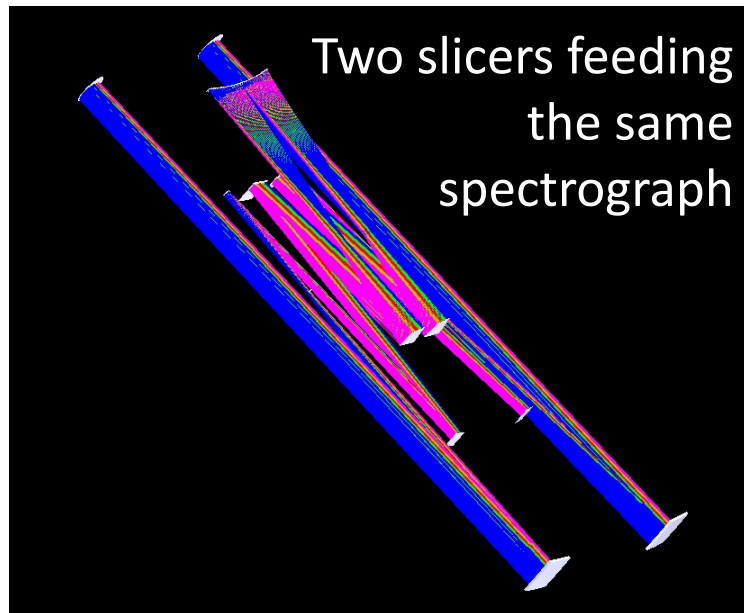
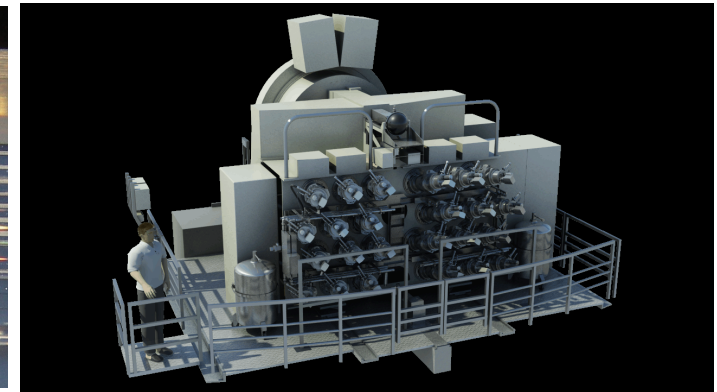
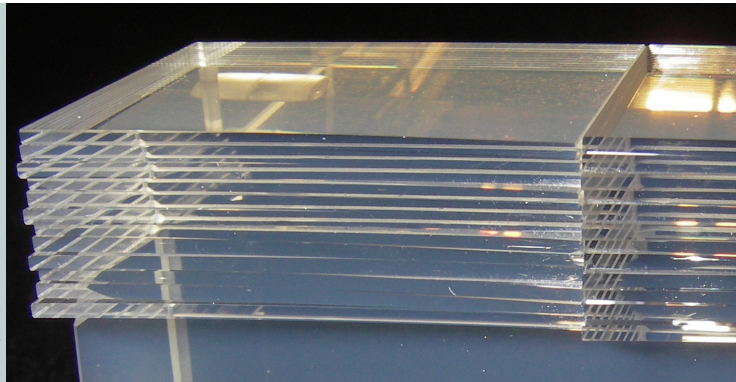
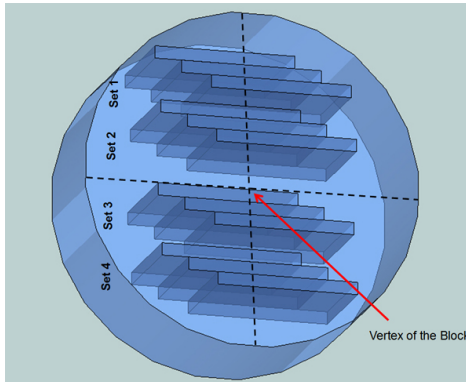
Tilt error of the slices  
 **$< 15$  arcsec PTV**

... applied on MUSE (2014) !

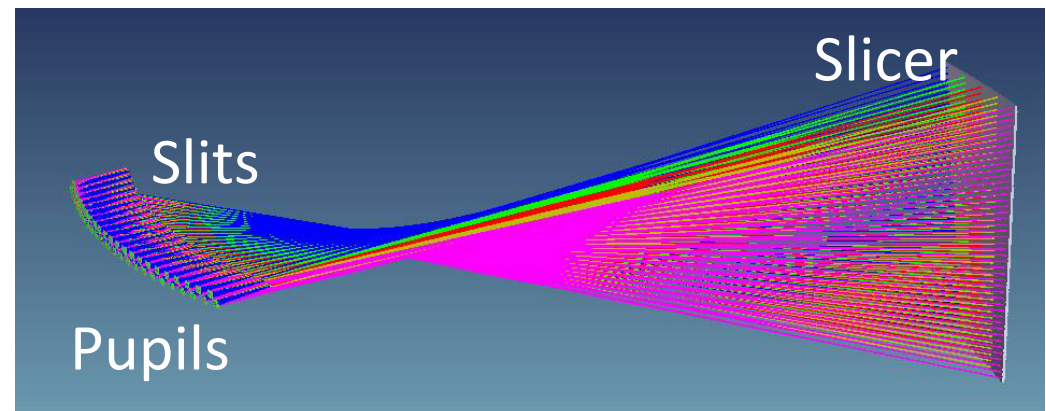


# Few optical designs we have proposed for ground based instruments

VLT/MUSE: Optimize manufacturing process

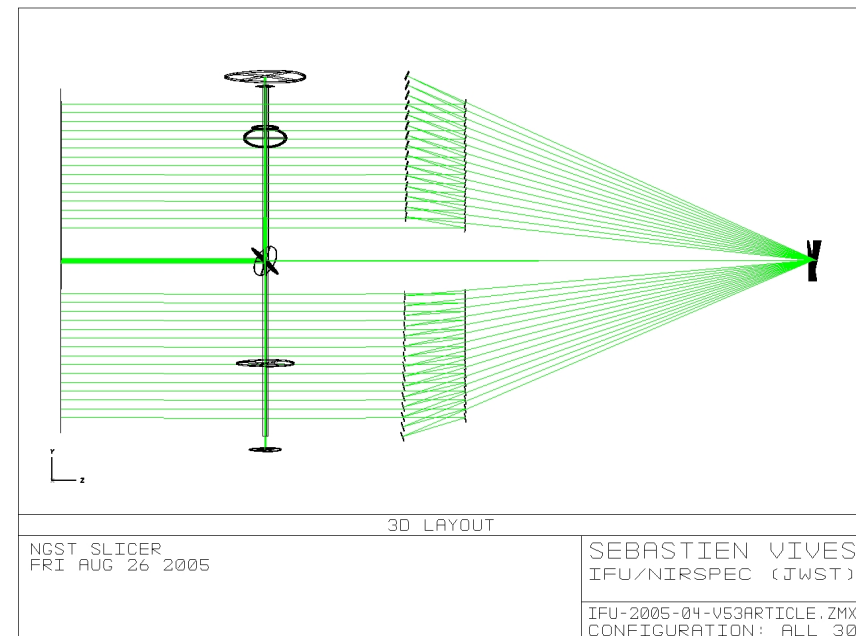
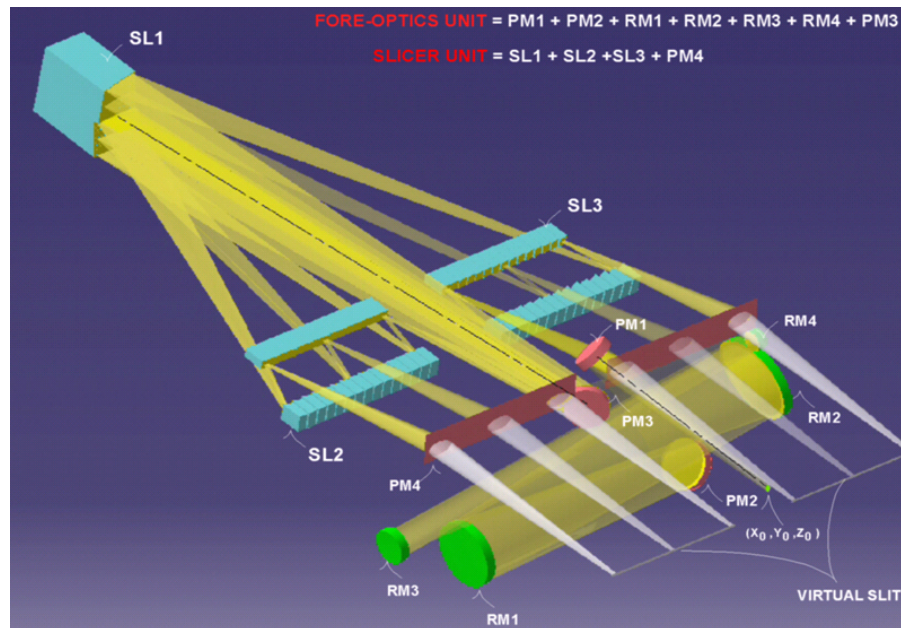


ELT/EAGLE:  
Minimize amount of optics



# Few optical designs we have proposed for spaceborn instruments

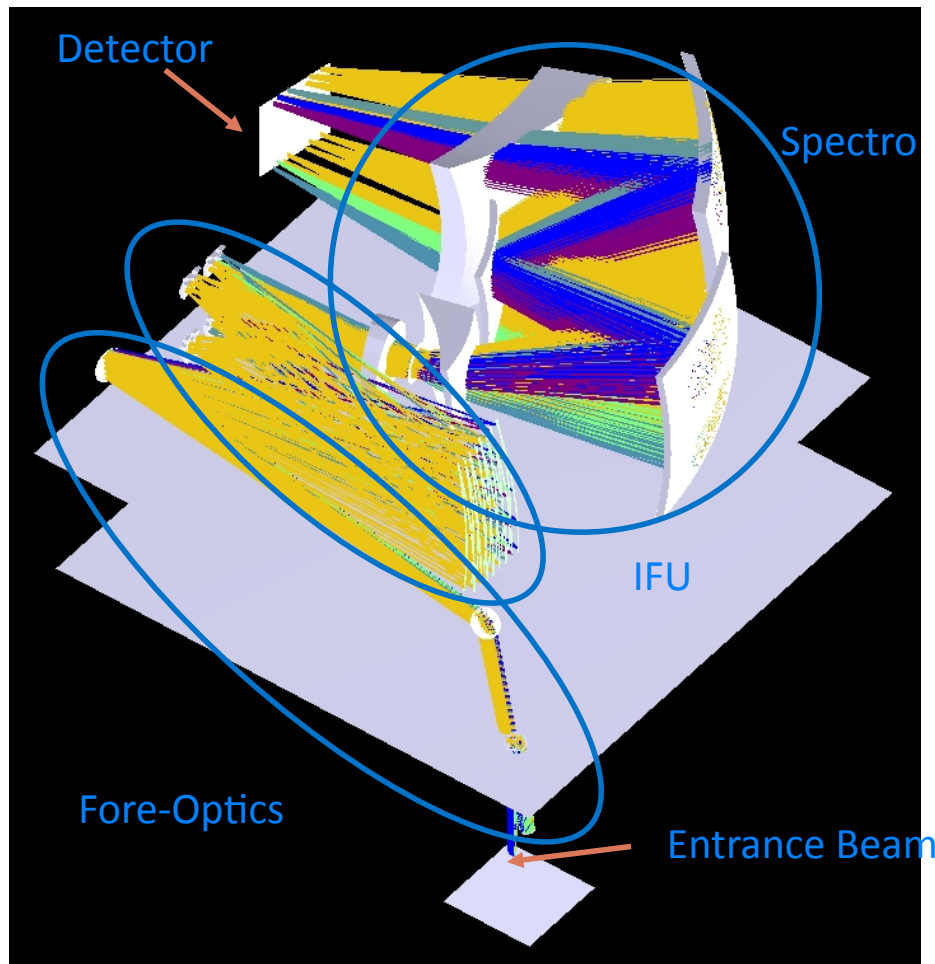
## JWST/NIRSpec: Optimize image quality





# Few optical designs we have proposed for spaceborn instruments

SNAP: Optimize image quality of both IFU+Spectrograph within limited volume



- ◉ Volume Allocation:
  - $\sim 270 \times 270 \times 200$  mm
- ◉ Field of view:
  - $10 \times 20$  arcsec<sup>2</sup>
- ◉ Wavelength range:
  - $0.4$  to  $1.7 \mu\text{m}$
- ◉ Spectral resolution:
  - $R \sim 100$  @  $1 \mu\text{m}$
- ◉ Spatial Sampling:
  - $0.15$  arcsec/px

## **The IFU technology is**

- ⊙ flexible and very powerful
- ⊙ well adapted to the WISH mission

**Some specifications have to be clarified (trades-off)  
before going more deeply into the optical design of the  
IFS**

# MERCI !

Et bon Appétit...